

NDSU AIAA

Design-Build-Fly
2021

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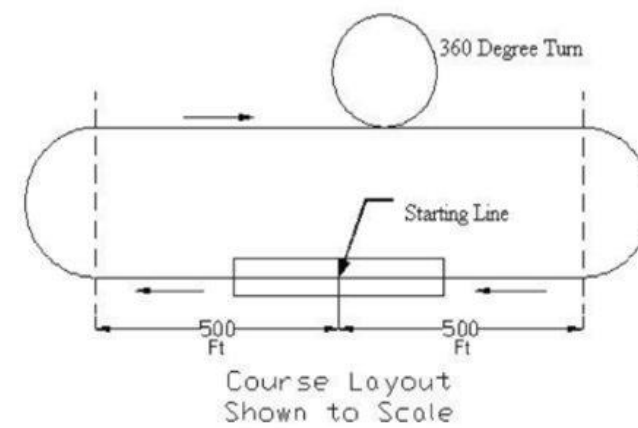
Noah Roth

AIAA Design-Build-Fly

- American Institute of Aeronautics and Astronautics Design, Build, Fly competition
- Goal: Design a UAV to successfully complete a set of specified missions
- This year's Competition: April 15-18 in Tucson, AZ



Mission Requirements

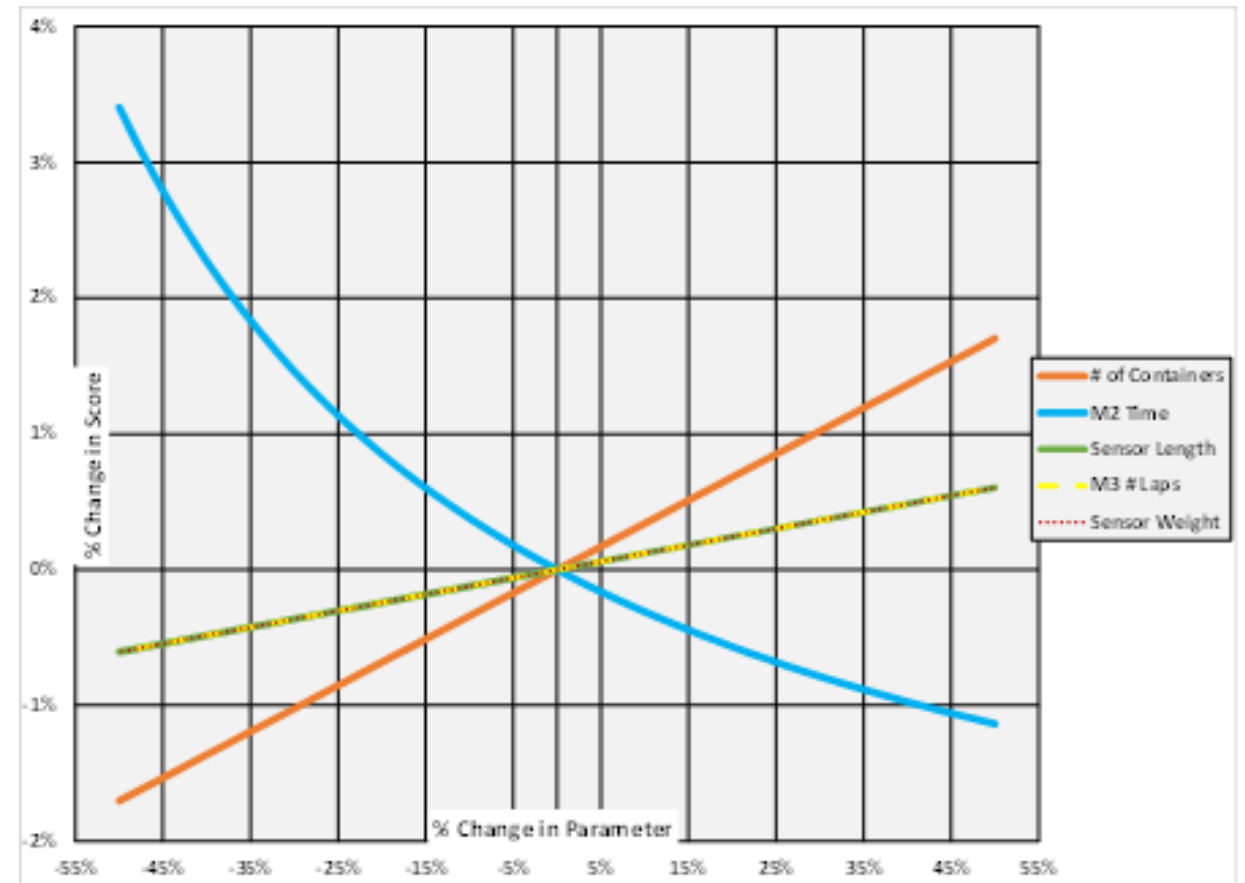


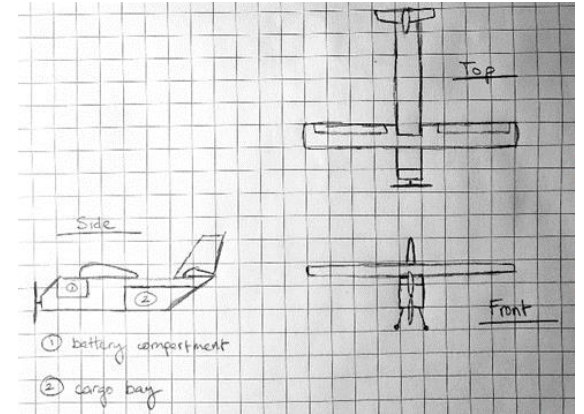
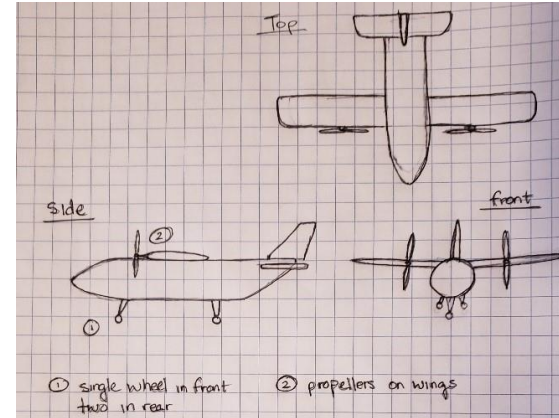
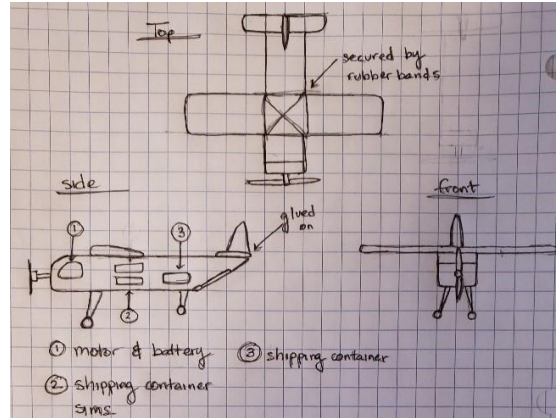
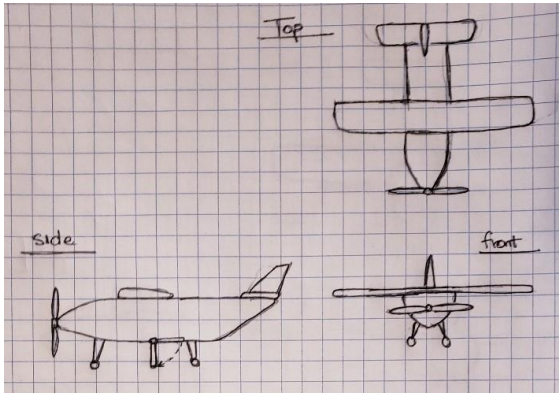
Missions	Mission Requirements/Scoring Criteria
Flight Mission 1: Staging Flight	5 minutes to complete 3 laps with no payload
Flight Mission 2: Delivery Flight	5 minutes to complete 3 laps with the sensor and simulator shipping containers
Flight Mission 3: Sensor Flight	10 minutes to deploy sensor and recover it before landing
Ground Mission	<ul style="list-style-type: none">• Demonstrate timely loading and unloading of shipping containers,• Demonstrate sensor deployment and recovery• Demonstrate all flight controls

Sensitivity Analysis

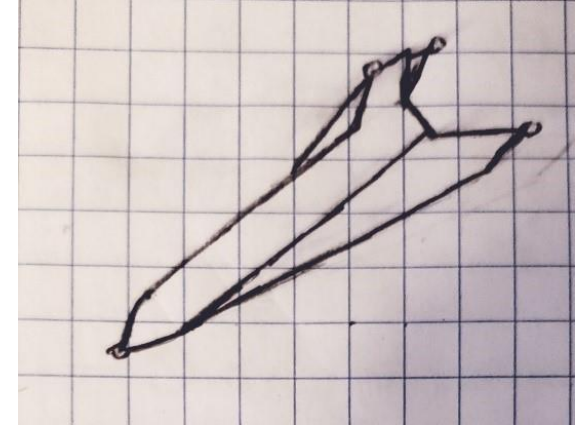
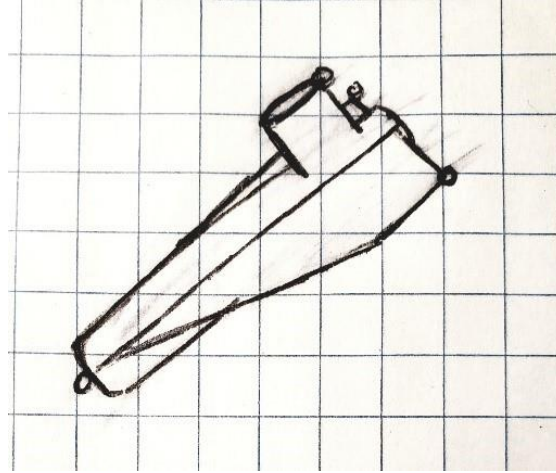
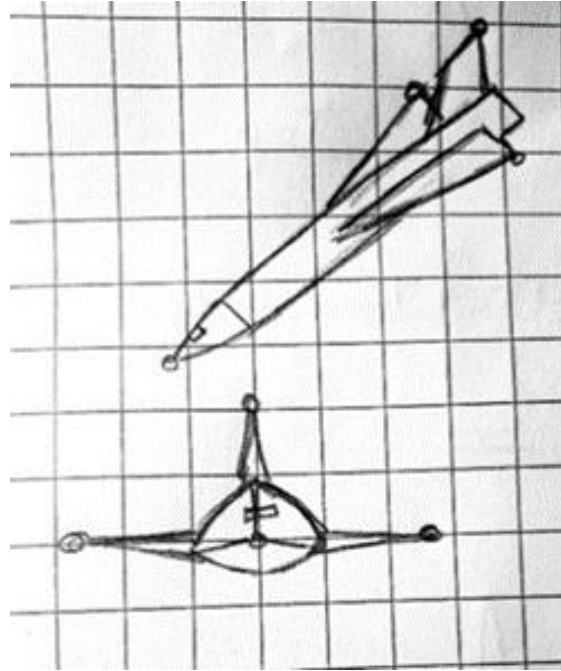
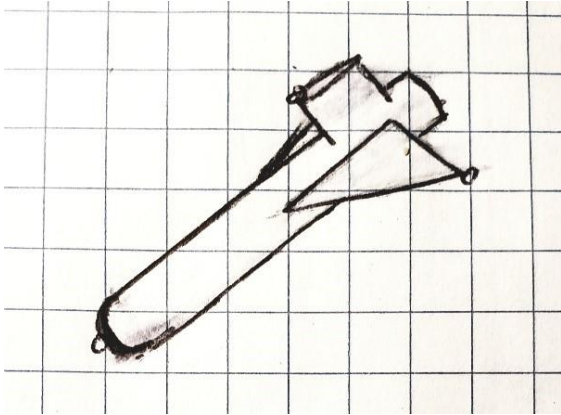
- Each mission scored with its own scoring equation
- Purpose: to evaluate effect of each scoring equation parameter
- Baseline parameters chosen based on initial designs
- Each increased and decreased by 2% up to 50% change
- Greatest percent change in score:
 - Mission 2 Time
 - # of shipping containers

	Scoring Equation
Flight Mission 1	$M1 = 1.0$
Flight Mission 2	$M2 = 1 + [N_{\text{(#containers/time)}} / \text{Max}_{\text{(#containers/time)}}]$
Flight Mission 3	$M3 = 2 + [N_{\text{(#laps * sensor length * sensor weight)}} / \text{Max}_{\text{(#laps * sensor length * sensor weight)}}]$
Ground Mission	$GM = [\text{Min}_{\text{time}} / N_{\text{time}}]$
Total Score	$M1 + M2 + M3 + GM + \text{Report Score}$

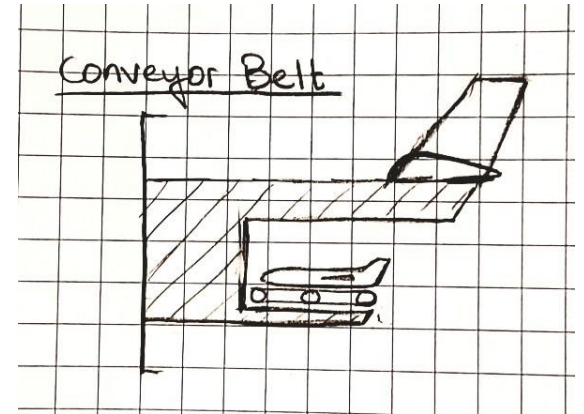
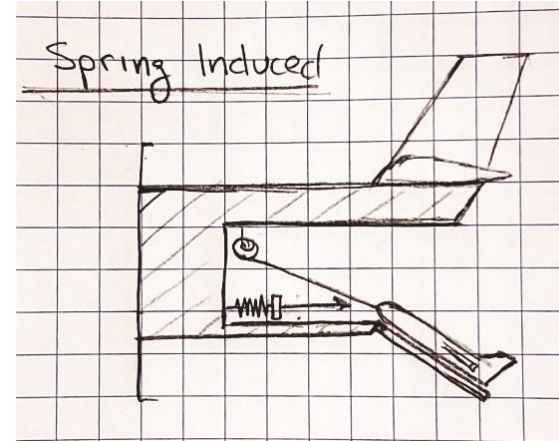
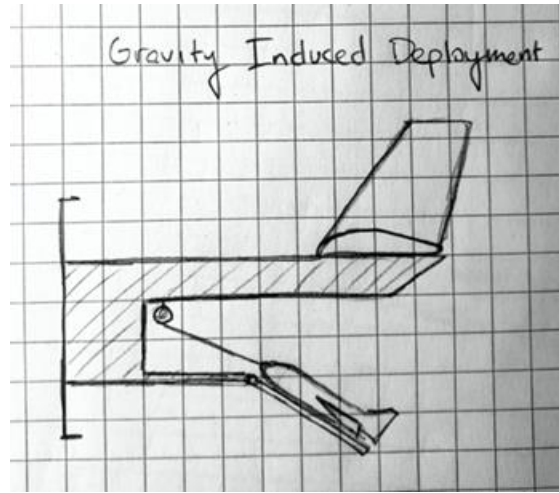
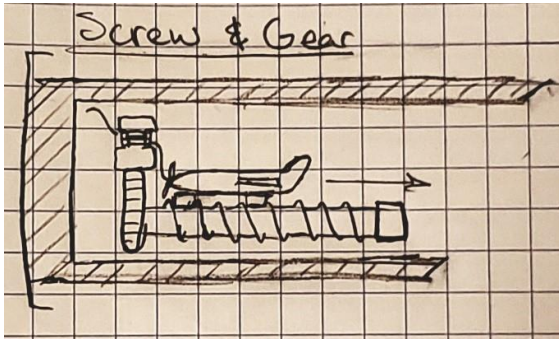




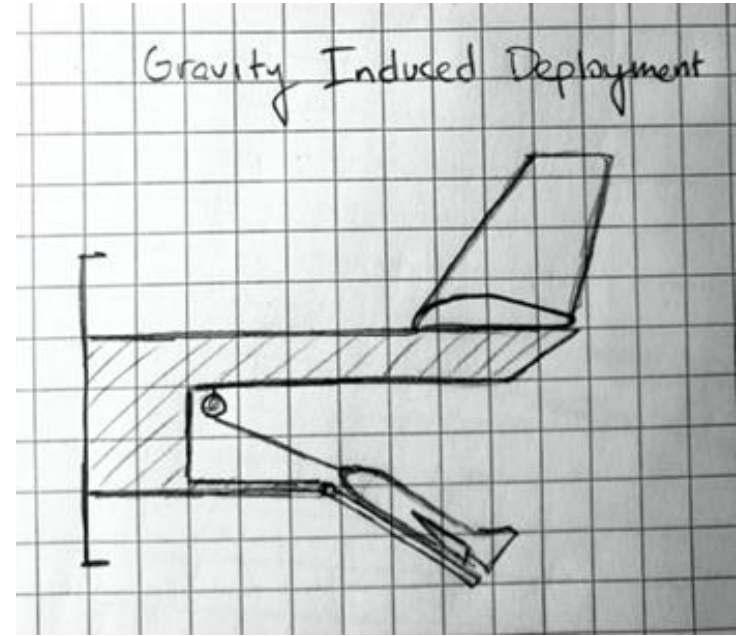
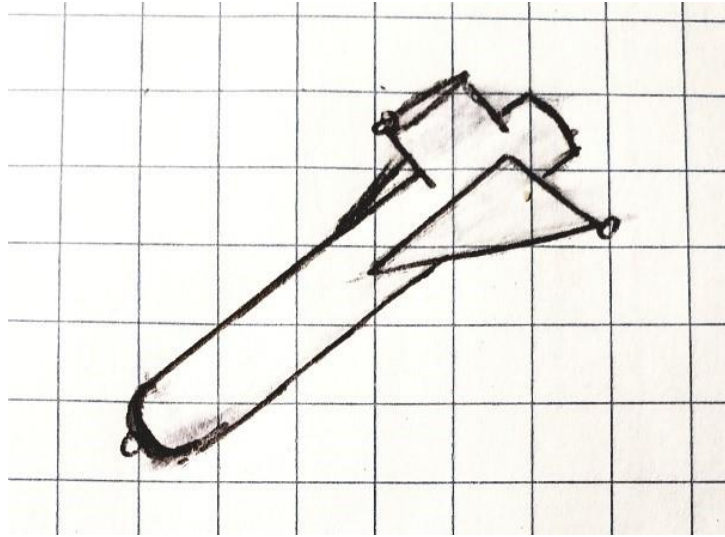
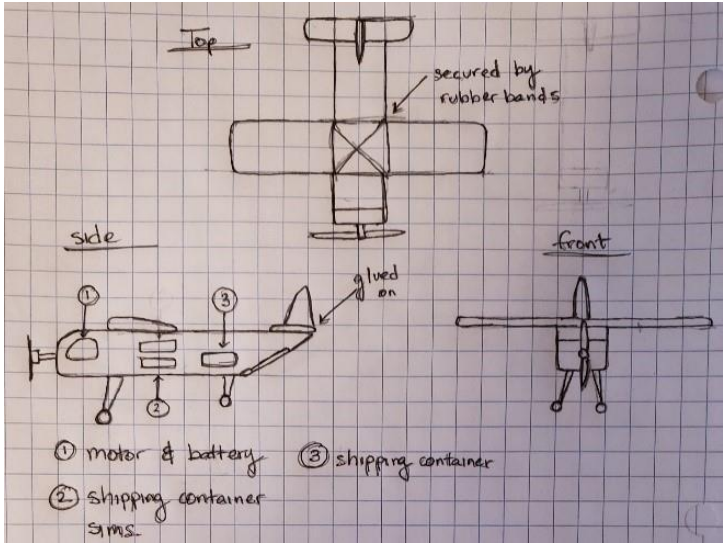
Concept Generation: UAV



Concept Generation: Shipping Container



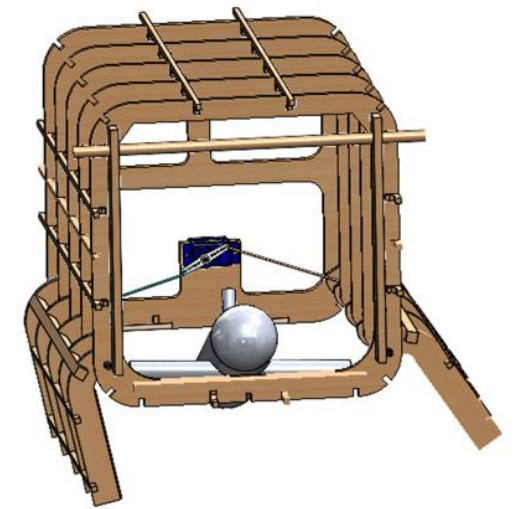
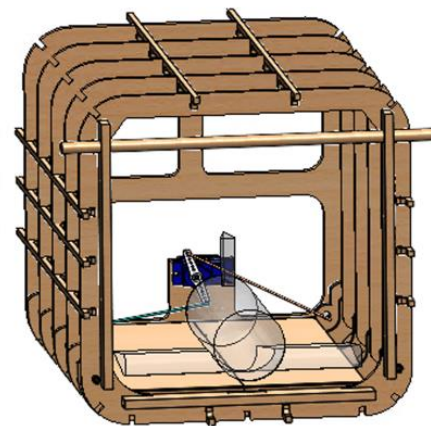
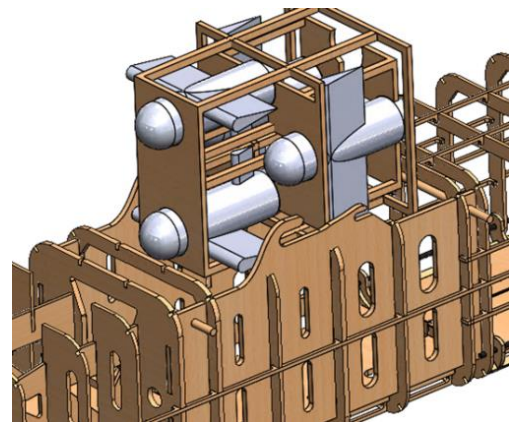
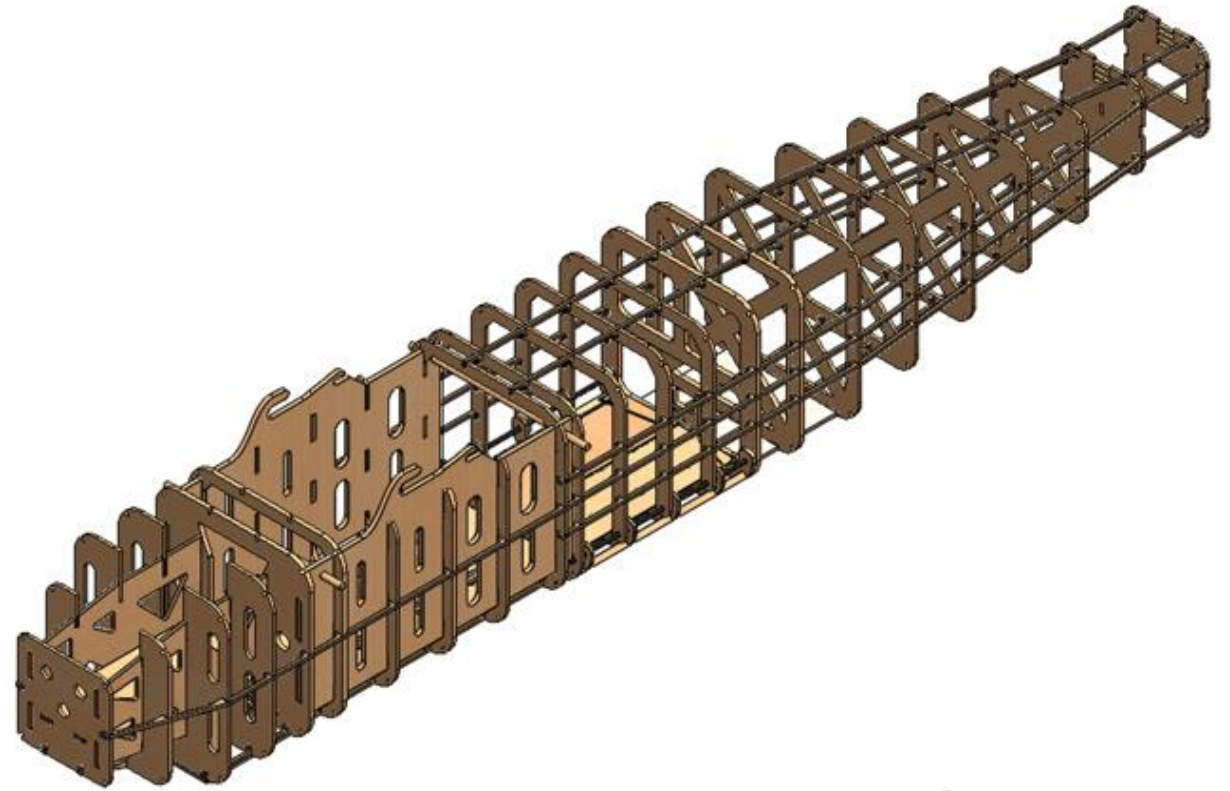
Concept Generation: Deployment Mechanism



Final Concepts

Design: Fuselage

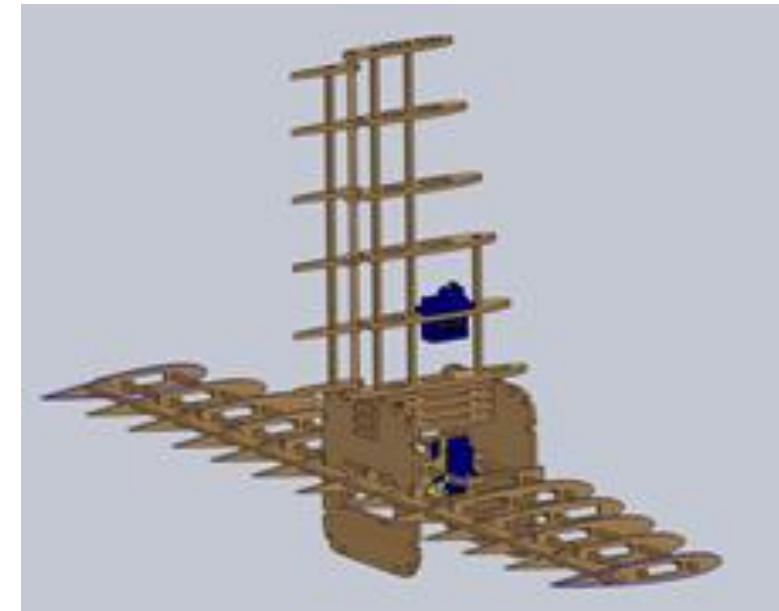
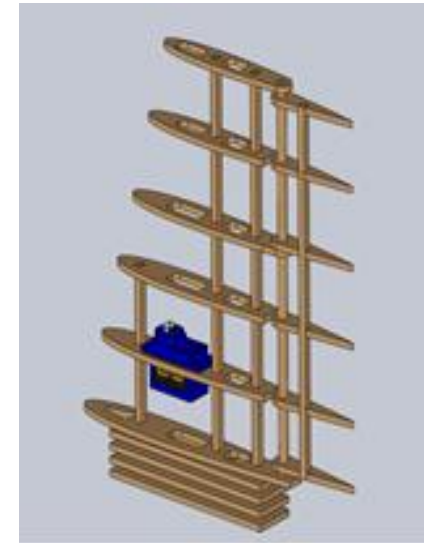
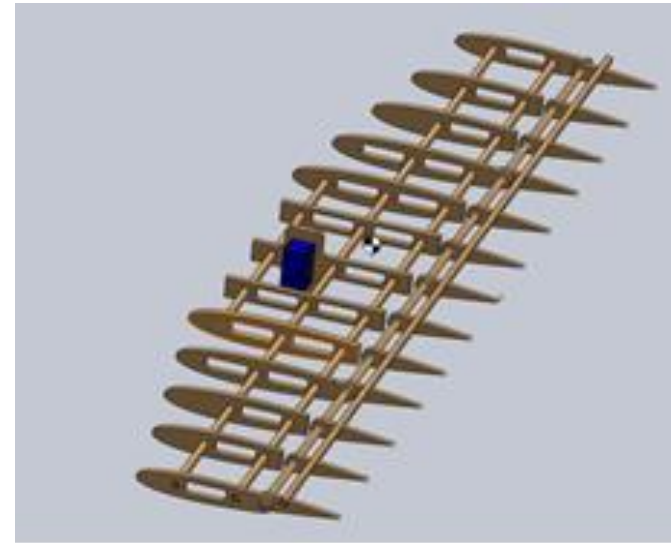
- Length: 5 feet
- 4 sections: Nose, cargo bay, bomb-bay, tail



Design: Tail

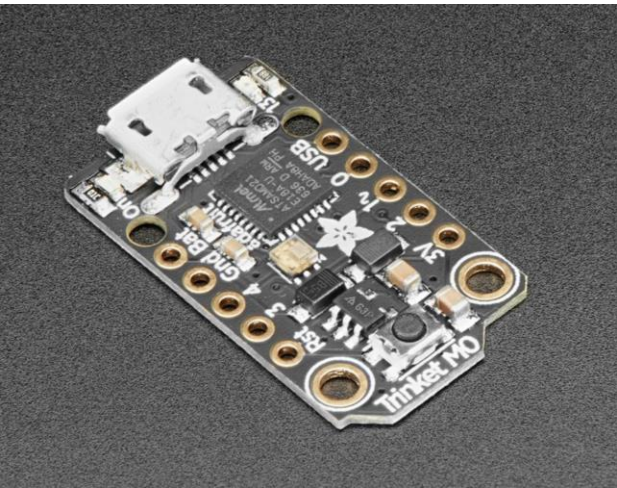
- Conventional tail
- Symmetrical airfoil (NACA0012)
- Calculations used to determine dimensions
- Horizontal and vertical stabilizers interlock with last two ribs of fuselage
- 0.125-inch-thick ribs
- 4mm carbon fiber rods used at the control surface hinges

	Horizontal Stabilizer	Vertical Stabilizer
Span (in)	20.875	7.5

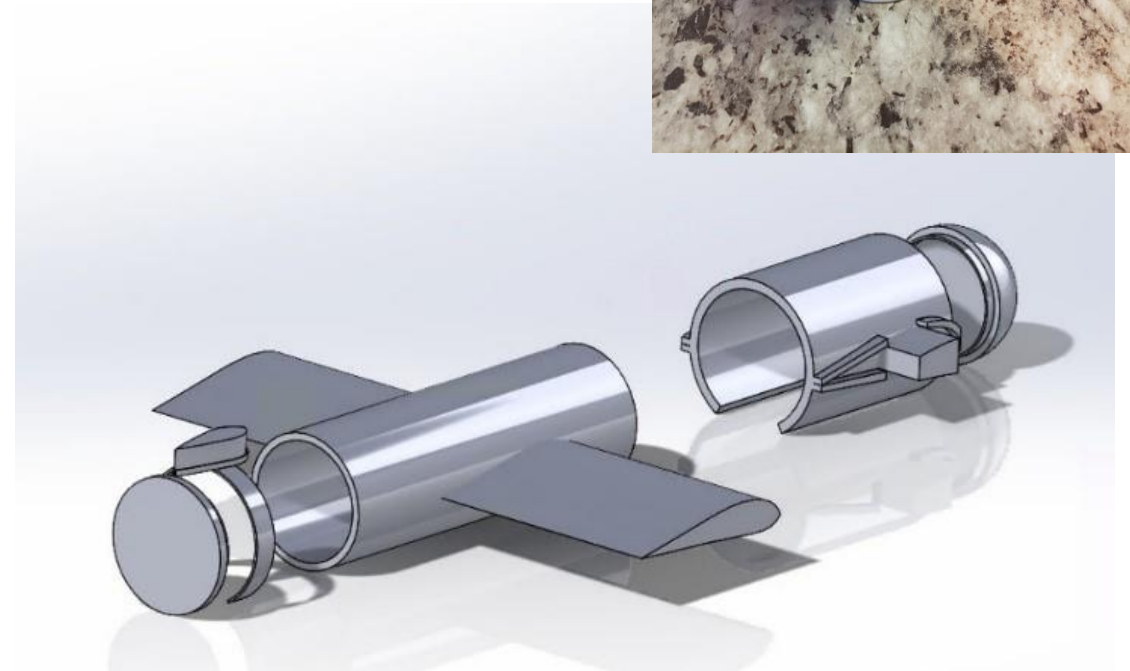


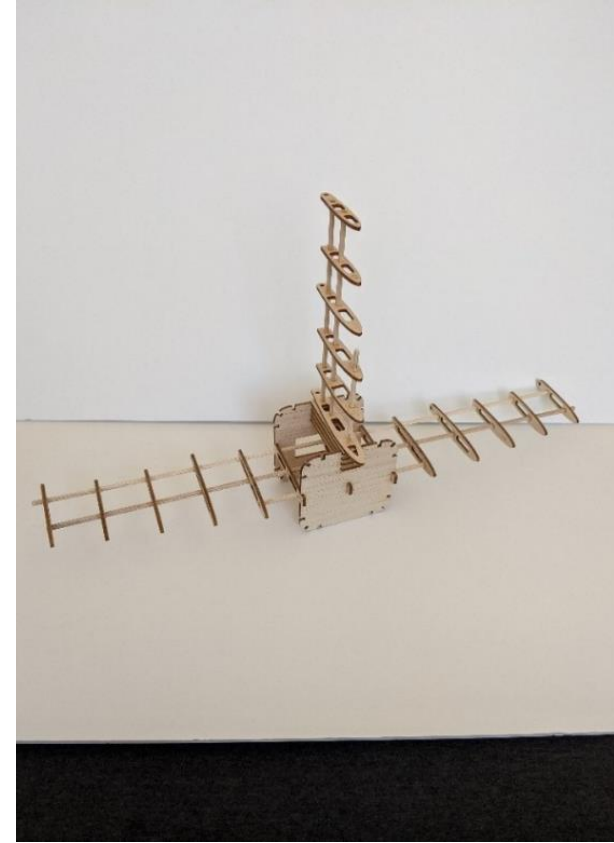
Design: Shipping Container

- Towable
- Holds the “sensor”



Trinket MO microcontroller



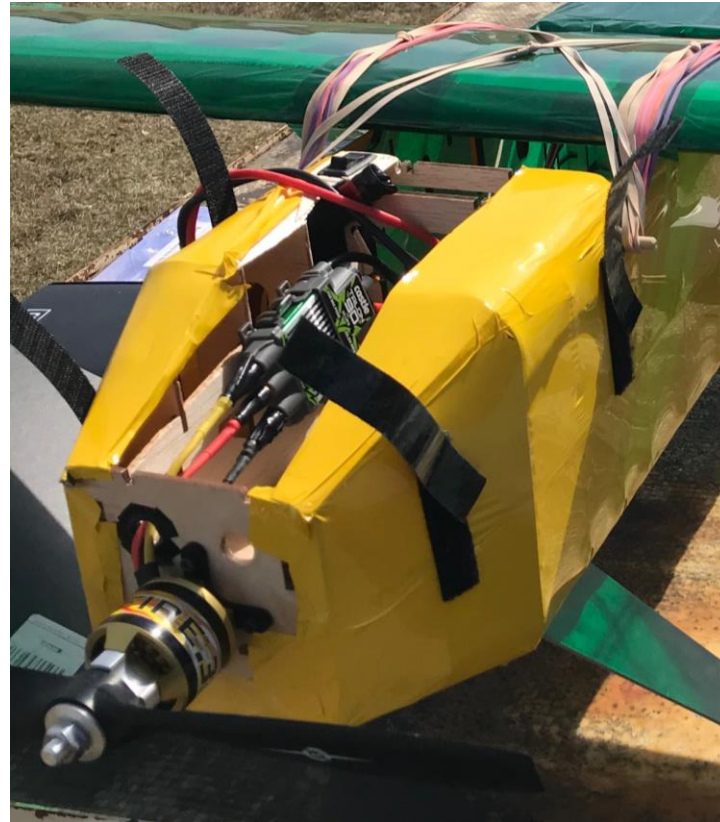


Manufacturing

- Built with balsa cut with a laser cutter
- Glued with super glue
- Covered with Monokote

Electronics

- Rimfire .32 electric motor
- 13x8 inch propeller
- 90 Amp ESC
- 3 and 4s LiPo batteries
- 8 channel Receiver
- 9 gram servos





Second Flight Test



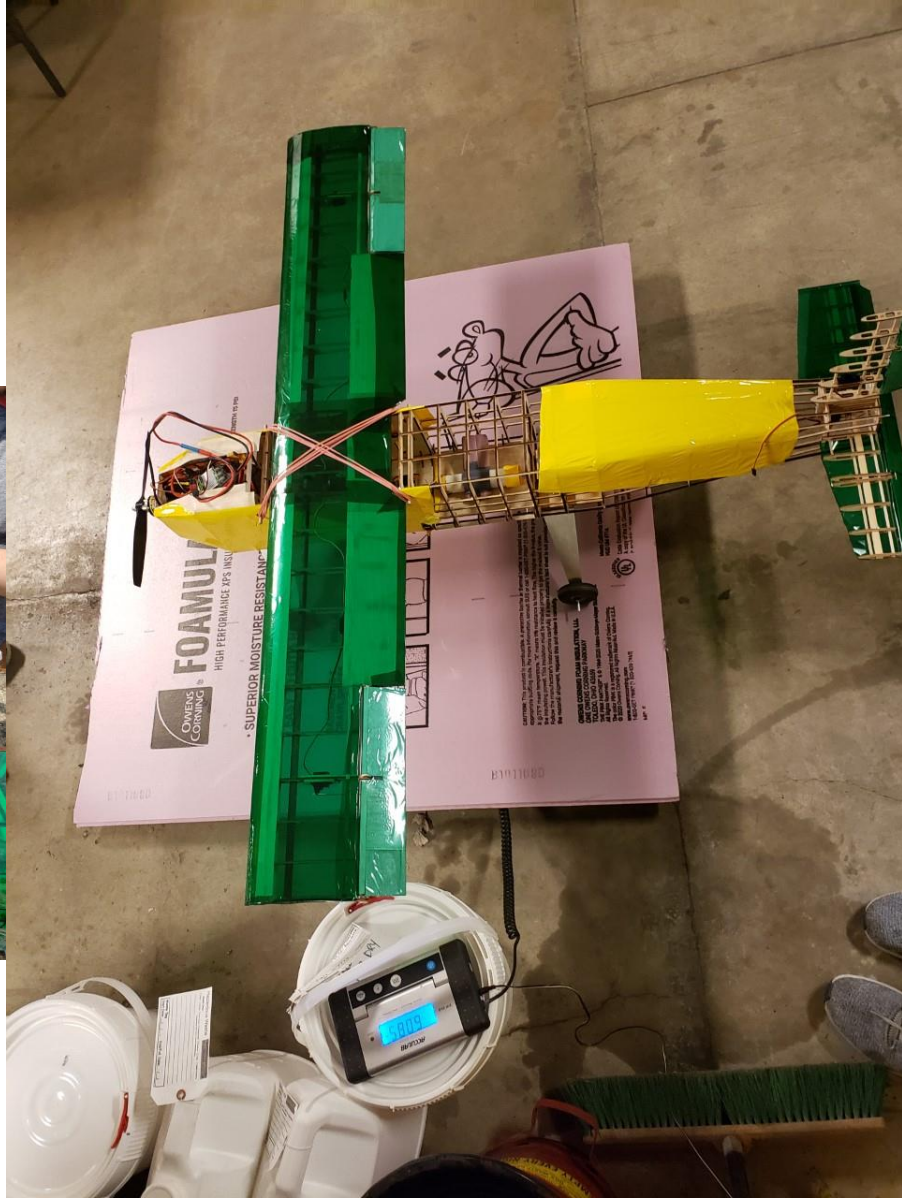
First Ground Test

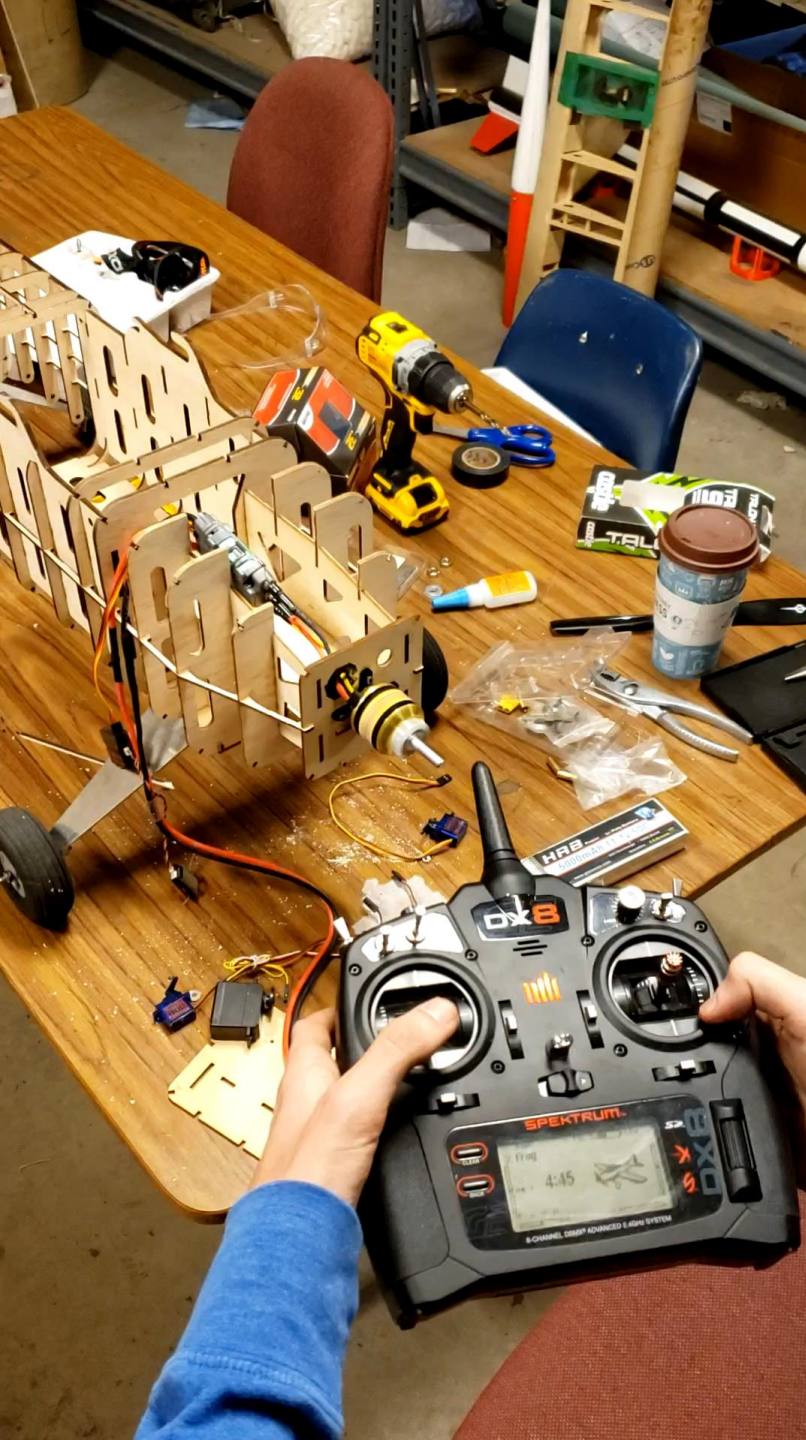


First flight test

Testing

- Tested electronics with ground tests
- Then brought it to a RC park to fly, got suggestions from the knowledgeable pilots
- Added a larger rudder than what we designed. Used a 4s battery for extra power.









Acknowledgments

- We'd like to thank the NDSU Mechanical engineering department for the opportunity to do this project
- NDSGC for their funding and support
- Our mentors Dr Suzen and Dr Zhang
- NDSU's AIAA Club



Questions?